

LOW VELOCITY IMPACT ANALYSIS ON COMPOSITE MATERIAL

C. Anand, K. Srinivasan, ChristuRajanM, Assistant Professor, Department of Mechanical Engineering
Dhanalakshmi Srinivasan College of Engineering and Technology, Mamallapuram

ABSTRACT:

This project aims at experimentally studying low velocity impact on fiber reinforced glass sheets. An experimental setup is fabricated, wherein a dart of known mass with accelerometer inbuilt in it made to fall on specimens of fiber reinforced glass sheets. The damage of the sheets of various thicknesses for various kinetic energies of dart is studied. The kinetic energy is varied by altering the height of the dart above the specimen sheet. The amount of energy transferred from the dart to the sheet is measured by the accelerometer output. The kinetic energy of the dart just before the moment it strikes the sheet is calculated by simple formula. The energy absorbed by the sheet is proportional to the thickness, material characteristics and the kinetic energy of the dart during collision. The dart is made fall free on the sheet; therefore, the acceleration is constant and is equal to the acceleration due to gravity constant (9.8 m sec^{-2}). Since the height of the dart above the sheet does not exceed three meters, the velocity during collision is comparatively low. A tabulation of velocities of dart, thickness of sheets and obtained. The accelerometer's electrical output is signal conditioned and is displayed in a computer using appropriate software. The goal of this project is to get a complete knowledge of the low velocity impact on the glass reinforced fiber sheets and to conclude how feasible it can be for various applications.

Keywords: Low velocity, Collision, kinetic energy, acceleration, voltage.
The experiments are redesigned to achieve a controlled plane-

1. INTRODUCTION:

Impact tests are used to study dynamic deformation and failure modes of materials. Low-velocity impact techniques can be classified as plate-on-plate, rod-on-plate, plate-on-rod, or rod-on-rod experiments. Low velocity impact material testing is significant because the defects or damages caused during such a collision are mostly unnoticed. This is a rod-on-plate type experiment. The response of structural composites to impact is studied. A composite material is a macroscopic combination of two or more distinct materials, having recognizable interface between them. Composite laminate is a combination of fiber and resin mixed in proper form. One of the unique properties of composite laminate is that it has high specific strength. Composites are being utilized as viable alternatives to metallic materials in structures where weight is a major consideration, e.g., aerospace structures, high speed boats and trains.

Plate impact experiments are used to generate such plane waves. These experiments provide controlled extreme stress-state loading conditions, involving one-dimensional stress-pulse propagation. The recovery configurations in plate-on-plate impact experiments are performed with the objective of examining the micro structural changes in the specimen after it is subjected to loading under a uniaxial strain condition. wave loading of the specimens. In practice, this is limited by the finite size of the plates employed, which generate radial release waves.

This has the potential for significant contribution to the damage processes by introducing causes other than the uniaxial straining of the material. Hence, this aspect of the plate impact experiment has been a subject of considerable research in the past. The plate impact experiments are performed in two main modes: normal impact and pressure-shear, or oblique, impact. Both modes have been specialized to several new configurations to achieve different aspects of control over the imposed loading. In these experiments, the time

histories of the stress waves are recorded and used to infer the response of the specimen with the goal of constitutive modeling. To enable the formulation of correct constitutive behavior for the considered material, knowledge of the micro mechanisms of deformation that occur during the passage of the stress waves is necessary. Such knowledge is also necessary for damage-evolution studies. Hence, it is important that the specimen is recovered after it is subjected to a well-characterized loading pulse so that it can be analyzed for any changes in its microstructure. This is achieved in the normal plate impact mode by using an impedance-matched momentum trap behind the specimen. Ideally, the momentum-trap plate captures the momentum of the loading pulse and flies away, leaving the specimen at rest.



Figure 1 Carbon Fiber Material

2. THE BASIC EXPERIMENTAL SETUP:

The experimental setup consists of a freefalling dart of known mass with accelerometer fixed into it and provision to make the dart collide head-on with the centre of specimen sheets. The accelerometer is wired and connected to a data acquisition system (DAS). The DAS gives it output to a computer with the appropriate software installed in it, which dynamically displays the acceleration in time output in X-Y axis.

The dart head arrangement

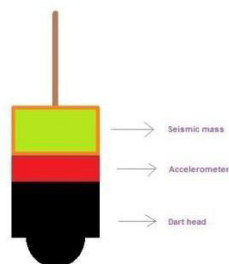


Figure 2: A Schematic Diagram of the Dart used for Collision

The dart arrangement consists of a dart head, which is the part that is going to collide head-on with the specimen sheet. The dart tip is therefore not sharp but made blunt to such an extent that there is no damage incurred on the sheet due to a pointed tip. The blunt area which is going to come in contact with the sheet is already measured. The energy which is going to be transferred to the sheet during the collision can be calculated only if the dimension and

the area of the region of contact are known.

The accelerometer is the most important part of this entire setup. The entire next chapter is dedicated to it. The accelerometer is sandwiched between the dart head and the seismic mass. It is the only direct measuring unit of the entire setup. It converts the mechanical energy into corresponding electrical output. It is robust and protected to handle well around the entire range of shock that can happen during the experiment.

The output and suspension mechanism the accelerometer has wired output. This wire is provided long enough so that the free fall of the dart setup is not affected at all. No drag is there due to the wire because of sufficient length. The seismic mass is located above the accelerometer. The purpose of the mass is to add weight to the dart setup. It is arranged in such a way that it does not have direct physical contact with the top surface of the accelerometer. The mass provides sufficient weight for the dart setup to have the necessary force to hit the specimen sheet due to free fall. The thread suspending the entire setup is strong. It is not rigid in shape because if it is rigid, it will affect the free fall. It is rolled in a pulley which is frictionless. The height of the fall of dart is adjustable by increasing and decreasing the entire pulley setup which is fixed moveably over a vertical stand of the experimental setup. A scale is provided on the vertical stand to measure the height from above which the dart is falling.

3. ACCELEROMETER:

Basics of accelerometer:

An accelerometer is a device that measures proper acceleration. This is not necessarily the same as the coordinate acceleration (change of velocity of the device in space), but is rather the type of acceleration associated with the phenomenon of weight experienced by a test mass that resides in the frame of reference of the accelerometer device. For an example of where these types of acceleration differ, an accelerometer will measure a value when sitting on the ground, because masses there have weights, even though they do not change velocity. However, an accelerometer in gravitational free fall toward the centre of the Earth will measure a value of zero because, even though its speed is increasing, it is in a frame of reference in which it is weightless. An accelerometer thus measures the change in acceleration. Another way of stating this is that by measuring weight, an accelerometer measures the acceleration of the free-fall reference frame (inertial reference frame) relative to itself (the accelerometer).

The accelerometer model:

The ADXL335 is a small, thin, low power, complete 3-axis accelerometer with signal conditioned voltage outputs. The product measures acceleration with a minimum full-scale range of ± 3 g. It can measure the static acceleration of gravity in tilt-sensing applications, as well as dynamic acceleration resulting from motion, shock, or vibration. The user selects the bandwidth of the accelerometer using the CX, CY, and CZ capacitors at the XOUT, YOUT, and ZOUT pins. Bandwidths can be selected to suit the application, with a range of 0.5 Hz to 1600 Hz for the X and Y axes, and a range of 0.5 Hz to 550 Hz for the Z axis.

The ADXL335 is available in a small, low profile, 4 mm \times 4 mm \times 1.45 mm, 16-lead, plastic lead frame chip scale package (LFCSP_LQ).

4. WORKING PRINCIPLE:

The basic working principle involved here is calculation of the energy absorbed by the specimen sheet by exploiting the concept of conservation of energy. When the dart hits the specimen, a part of the kinetic

energy of the dart is transferred to the specimen. The specimen breaks and the dart penetrates it. The kinetic energy of the dart after the breaking of the specimen is equivalent to the difference

between the kinetic energy it had just before the collision and the energy that was absorbed by the specimen.

The below equation is governing equation to calculate the energy that is being imparted to the specimen. $E_{bc} = E_{ac} + E_{as}$

E_{bc} - Energy of dart before collision

E_{ac} - Energy of dart after collision

E_{as} - Energy absorbed by the specimen

$$E_{bc} = m * g * h$$

E_{ac} is calculated as half of the product of mass and the square of the velocity.

$$E_{ac} = (1/2) * m * v^2$$

m - mass of the dart

g - acceleration due to gravity

h - height of the dart above the ground

v - velocity of the dart just after the collision E_{as} is obtained by subtracting E_{ac} from E_{bc} .

The velocity "v" needed to calculate E_{ac} is obtained by the acceleration output obtained from the dart. When it is integrated with respect to time once, velocity is obtained.

The voltage output of the accelerometer is linearly proportional to acceleration. The integrating software is MATLAB. The voltage output obtained is plotted continuously with time and a graph is obtained. The velocity is to time output is obtained by performing a single point to point integration of the values in the previous mentioned graph.

The output from the DAQ is obtained by a simple RS 232 connection. The baud rate and other specifications prerequisite to receive the data are done and a simple MATLAB code is written to receive the voltage data. The MATLAB programme is presented below:

```
sa=serial('COM9','BaudRate',9600);fopen(sa);
b=[];
%sum=0;
%d(0)=0;i=10;
forj=1:i
    pause(0.5);tic ;
a=fgetl(sa);b=a;disp(b);time=toc;disp(time);
%d(j)=time+sum;
%sum=d(j);endfclose(sa);clear('sa');
%plot(d,c);
```

The baud rate can be adjusted for different kind of DAQs. The COM port may differ from PC to PC. The time interval for which the DAQ should continuously acquire the data is programmed. Precaution must be taken to see that the time interval to acquire data is in synchronisation with the time taken to transmit the data. If the time interval of acquisition is considerably smaller than the time taken for the RS 232 system to transmit the data to the PC, then a number of readings of the data will be missed or the system will throw an error to show the timings are not in sync.

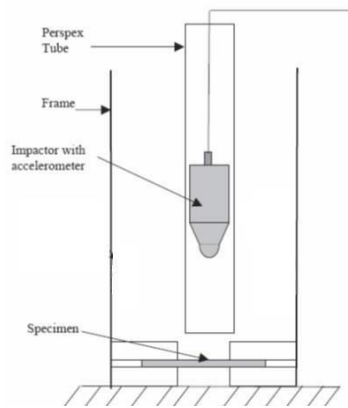


Figure 3A Schematic Diagram of the Dart used for Collision

5. REFERENCES:

- [1] **Ceyla Akin and Mehmet Şenel**, “An Experimental Study of Low Velocity Impact Response for Composite Laminated Plates”, DPÜ Fen Bilimleri Enstitüsü Dergisi, Sayı 21, Nisan, 2010.
- [2] **Constantinides G., Tweedie C.A., Savva N., and Smith J.F., Van Vliet K.J.**, “Quantitative Impact Testing of Energy Dissipation at Surfaces Experimental Mechanics”, 49:511–522, 2009.
- [3] **Davies G.A.O. and Olsson R., Imperial College London**, “Impact on composite Structures”, The Aeronautical Journal, 541–563, 2004.
- [4] **Rajesh Mathivanan and N., Jerald J.**, “Experimental Investigation of Woven E-Glass Epoxy Composite Laminates, subjected to Low-Velocity Impact at Different Energy Levels”, J. Minerals & Materials Characterization & Engineering, Vol. 9, No. 7, pp. 643-652, 2010.
- [5] **Rajkumar. G.R.**, “Effect of Low Velocity Repeated Impacts on Property Degradation of Aluminum-Glass Fiber Laminates”, International Journal of Engineering Science and Technology, Vol. 3 No. 5, pp 4131-4140, 2011.